

## SUPPLEMENTAL TYPE CERTIFICATE

# 10066697

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

# AVIONICS SUPPORT GROUP, Inc.

13155 SW 132 AVENUE **MIAMI FL 33186** USA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.A.205

Type Certificate Holder: THE BOEING COMPANY

**Type: 757** 

Model: 757-200/-300

Original STC Number: FAA STC ST03539AT

### Description of Design Change:

Installation of a Thales Integrated Standby Flight Display in B757-200/-300 Series Aircraft.

#### **EASA Certification Basis:**

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

#### **Associated Technical Documentation:**

1. Avionics Support Group, Inc. Master Data List 27252-MDL, Rev. IR, dated 14 November 2007, or later FAA approved revision.

See Continuation Sheet(s)

For the European Aviation Safety Agency Cologne, Germany, 30 August 2018

> Dominique RENAUD **Acting Business Aeroplanes Section Manager**





2. Airplane Flight Manual Supplement for Avionics Support Group 27252-FMS dated 26 February, or later FAA approved revision.

## Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

