

SUPPLEMENTAL TYPE CERTIFICATE

10063553

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVIONICS SUPPORT GROUP, Inc.

13155 SW 132 AVENUE
MIAMI FL 33186
USA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.120

Type Certificate Holder: THE BOEING COMPANY

Type: 737

Model: 737-600/-700/-800/-900

Original STC Number: FAA STC ST04408AT

Description of Design Change:

Installation of Rockwell Collins TPR-901 Automatic Dependent Surveillance - Broadcast (ADS-B) Out Transponder(s).

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

Regulations at the latest amendment 25-0 through 25-143:

§25.29, §25.771, §25.773, §25.777, §25.1301(a), §25.1307(c), §25.1309, § 25.1322, §25.1351(a), §25.1353(a), §25.1357(a), §25.1360, §25.1431(a)(c), §25.1529, §25.1541, §25.1581.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 23 October 2017



Colin HANCOCK
Section Manager
Regional Transport Aeroplanes

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

1. Master Data List (MDL) 216307-MDL, Rev. B, dated 17 August 2017, or later FAA approved revision.
2. Instructions for Continued Airworthiness (ICA) Document 216307-ICA, Rev. A, dated 19 May 2017, or later FAA accepted revision.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -