

SUPPLEMENTAL TYPE CERTIFICATE

10068730

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVIONICS SUPPORT GROUP, Inc.

**13155 SW 132 AVENUE
MIAMI FL 33186
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.A.004

Type Certificate Holder: AIRBUS S.A.S.

Type: A330

Model: A330-201, A330-202, A330-203

A330-223, A330-223F

A330-243, A330-243F

A330-301, A330-302, A330-303

A330-321, A330-322, A330-323

A330-341, A330-342, A330-343

Original STC Number: FAA STC ST04247AT

Description of Design Change:

Installation of an Avionics Support Group Constant Friction Mount (cfMount) and Cradle with an Integrated Power Supply (IPS) on Airbus A330 Series Aircraft.

EASA Certification Basis:

The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC ST04247AT.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 13 February 2019


Carla IORIO
Supplemental Type Certificates & Special Projects
Section Manager



The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

1. Avionics Support Group, Inc. Master Data List 214255-MDL, Rev. H, dated 29 November 2016, or later FAA approved revision.
2. Instructions for Continued Airworthiness Document 214255-ICA dated 21 November 2016 Rev. F or later approved revision.
3. Airplane Flight Manual Supplement 215289-FMS is a required part of this STC.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -