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Numberes sto4147AT

This certificate issued to

Avionics Support Group 13155 SW $132^{\text {nd }}$ Avenue Miami, FL 33186

sertifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airucrthines's requirements of Part 25 of the Federal. Avviation Regulations.

Originat SProduct-TYpne Cerlificate Nember:

A57NM
Embraer
190-100/-200

Descriztion of Type Design Change: Installation of a Constant Friction Mount and Cradle on Embraer 190-100/200 series aircraft and associated hardware in accordance with Master Drawing List 212034-MDL, Revision C, dated February 20, 2014 or later FAA approved drawing revisions.

Cimilations and Condilions: This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated, unless it is determine by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect upon the airworthiness of that airplane. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

Limitations and conditions continued on page 3 of 3

[^0]Date of application: June 15, 2013
Date of issuance: March 25, 2014


Date reissued:
Date amended:
By direction of the todministratore
(Signature)
Melvin D. Taylor
Manager,
Atlanta Aircraft Certification Office

Instructions for Continued Airworthiness (ICA), Avionics Support Group Document No. 212034-ICA, Rev. A, dated November 15, 2013, or later FAA accepted revision must be made available to the operator at the time of installation.

Based on 14 CFR $\S \S 21.115$ and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for Embraer ERJ 190-100/-200 aircraft is as follows:
a. The type certification basis Embraer 190-100/-200 aircraft is shown on TCDS A57NM for parts not changed or not affected by the change.
b. The certification basis for parts changed or affected by the change since the reference date of application, June 15, 2013 is based upon Part 25 as amended by Amendment 25-136. Based on 14 CFR §§ 21.115 and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for this modification was determined to be:

Regulations at the latest amendment 25-0 through 25-136
$25.29,25.301,25.303,25.305$ (a)(b) $1,25.307$ (a), $25.561,25.601,25.603,25.605 .25 .609,25.611,25.613$, 25.789(a), 25.1301(a) $23,25.1307$ (c), 25.1309, 25.1529, 25.1541

## Regulations at an intermediate amendment

none

## Regulations at the amendment level in TCDS A57NM

## 14 CFR Part 26 Regulations

Based on 14 CFR $\$ 21.101(\mathrm{~g})$, applicable provision of 14 CFR part 26 are included in the certification basis. For any future 14 CFR part 26 amendments, the holder of this STC must demonstrate compliance with the applicable sections. As of the issuance date of this STC, compliance has been found for the following regulation: 14 CFR §26.11.


[^0]:    This serlificate and the supporting data which is the basid for approval ishall remain in offect until surrendered, suspended. revolved or a termination date is otherwise established by the Aodministrator of the Federat Aoviation Aodministration.

