European Aviation Safety Agency



SUPPLEMENTAL TYPE CERTIFICATE

10048424

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVIONICS SUPPORT GROUP, Inc.

13155 SW 132 AVENUE MIAMI FL 33186 USA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number : EASA IM.A.120 Type Certificate Holder : THE BOEING COMPANY Type Design - Model : 737-300/-400/-500 Type Design - Model : 737-600/-700/-800 Original STC Number : FAA ST03081AT

Description of Design Change:

Installation of an emergency Locator Transmitter (ELT) system.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 10 March 2014

9 Michel

Slawa OTITSCH Project Certification Manager Large Aeroplanes

Note:

The following numbers are listed on the certificate: EASA current Project Number: 0010028162-001

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Associated Technical Documentation:

Master Drawing List 24694-MDL Rev B dated 04/04/2007 Flight Manual Supplement 24694-AFMS initial issue or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

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